

# PROJECT facts

U.S. DEPARTMENT OF ENERGY  
OFFICE OF FOSSIL ENERGY  
NATIONAL ENERGY TECHNOLOGY LABORATORY

High Efficiency  
Engines and Turbines

12/2002



## EXTENDING THE LEAN BLOWOUT LIMITS OF LOW NO<sub>x</sub> GAS TURBINES BY CONTROL OF COMBUSTION INSTABILITIES

### Description

#### PRIMARY PROJECT PARTNER

Georgia Institute of Technology  
Atlanta, GA 30332

#### TOTAL ESTIMATED COST

\$ 510,329.

#### CUSTOMER SERVICE

800-553-7681

#### STRATEGIC CENTER FOR NATURAL GAS WEBSITE

[www.netl.doe.gov/scng](http://www.netl.doe.gov/scng)

This Advanced Gas Turbine Systems Research (AGTSR) project builds on the success of an earlier AGTSR project that identified a promising active control approach for low NO<sub>x</sub> turbine combustors. That control scheme periodically modulates a portion of fuel injected into the combustor to produce heat release oscillations out of phase with the combustor oscillation. The approach was successfully tested at a Siemens-Westinghouse combustor facility. The following figure shows pressure oscillations in a test combustor during operation without control, during an identification period when the oscillations are monitored to set control parameters, and after control has been initiated. The active control has produced a four-fold reduction in amplitudes of oscillations.

Two patents have been issued and a third is in process for this active control approach. GT has been transferring this technology to the US gas turbine industry. As a result of its success, GT is establishing working arrangements with several gas turbine manufacturers to test or evaluate adapting this control approach to turbine combustors. These companies include General Electric, Rolls-Royce, Siemens-Westinghouse, and United Technologies Research Center (Pratt&Whitney).

One of the three tasks of the current AGTSR program will further improve and produce a practical active control system based on the approach developed in the previous AGTSR program. Two additional tasks are investigating mechanisms and passive control by i) distributed fuel injection, ii) modification of fuel line properties to eliminate or reduce coupling between fuel line and combustor inlet oscillations, iii) pulsation of the pilot fuel injection rate at a fixed frequency, iv) addition of a tunable Helmholtz resonator at the inlet to increase acoustic damping, and v) use of a splitter pipe to produce periodic flow vortex structures that are out of phase.



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## CONTACT POINTS

### Tom George

National Energy Technology  
Laboratory  
P.O. Box 880  
3610 Collins Ferry Rd  
Morgantown, WV 26507-0880  
304-285-4825  
tgeorge@netl.doe.gov

### Richard Dennis

Product Manager,  
High Efficiency Engines and  
Turbines  
National Energy Technology  
Laboratory  
P.O. Box 880  
3610 Collins Ferry Rd.  
Morgantown, WV 26507-0880  
304-285-4515  
richard.dennis@netl.doe.gov

### Richard Wenglarz

South Carolina Institute for  
Energy Studies  
386-2 College Ave.  
Clemson, SC 29634  
864-656-2267  
rwnglrz@clemson.edu

### Professor Ben Zinn

Georgia Institute of Technology  
270 Ferst Drive  
Atlanta, GA 30332  
404-894-3033  
ben.zinn@aerospace.gatech.edu

## Duration

36 months

## Project Goals

This program investigates both passive and active control of detrimental combustion instabilities in low NO<sub>x</sub> turbines, which burn natural gas in a lean premixed mode to reduce emissions. Methods of passive control of instabilities are being explored because of simplicity of implementation. Active control of instabilities is being explored because adequate passive control may not be possible for all low emission turbines.

## Project Benefits

Lean premixed low NO<sub>x</sub> turbine combustors operate in a regime closer to lean blowout limits than conventional combustors. This results in combustion pressure oscillations and instabilities with accompanying noise and vibration. Such combustor noise has been unacceptable and vibration induced fatigue has caused structural failures in engines and removal of commercial turbines from service for repair. This project will develop control approaches to alleviate combustor oscillations, noise and damaging vibrations. These control approaches will also offer the potential for operation of low emission combustors closer to the lean blowout limits where additional reduction of NO<sub>x</sub> emissions is possible.

